

## NONDIMENSIONAL PARAMETER ANALYSIS

The maximum pressure a container will withstand is a function of the material fatigue strength, the amount of prestress, the number of components  $N$ , and the wall ratios  $k_n$ . To determine the function dependence on these variables and to determine the best designs, a nondimensional analysis is now presented. The calculations for the analysis of each design were programmed on Battelle's CDC 3400 computer.

### Multiring Container

#### Static Shear Strength Analysis

Although a fatigue criterion of failure has been chosen it is illustrative to review an analysis based upon static shear strength for ductile materials first conducted by Manning<sup>(23)</sup>. The method outlined here differs from that of Manning and is more straightforward. In this analysis the optimum design is found such that each component of the same material has the same value of maximum shear stress  $S$  under the pressure load  $p$ . The given information is  $p_0 = p$ ,  $p_N = 0$ , and  $K$ . The unknowns are the interface pressures  $p_n$ ,  $(N-1)$  in number; the  $k_n$ ,  $N$  in number and  $S$ . The total unknowns are  $2N$ . There are  $N$  equations resulting from Equation (15) and having the form

$$S = (p_{n-1} - p_n) \frac{k_n^2}{k_n^2 - 1}, \quad n = 1, 2, \dots, N \quad (26)$$

There is the equation,  $K = k_1 k_2 \dots k_n$ , that relates the  $k_n$  and  $K$ . Also  $N-1$  equations can be formulated from the requirement that  $S$  be a minimum, i. e.,

$$\frac{\partial S}{\partial k_n} = 0, \quad n = 1, 2, \dots, N-1 \quad (27)$$

(There are not  $N$  equations in the Form (27) because there is one equation relating the  $k_n$ .) Thus, there are also  $2N$  equations which can be solved for the  $2N$  unknowns. The solution gives

$$p_n = p_{n-1} - \frac{(k_n^2 - 1)}{k_n^2} S, \quad n = 1, 2, \dots, N-1 \quad (28)$$

$$k_1 = k_2 = \dots = k_N \quad (29)$$

$$S = \frac{p}{N} \frac{K^{2/N}}{(K^{2/N-1})} \quad (30)$$

The residual pressures  $q_n$  and the required interferences for the shrink-fit assembly have yet to be found. The radial stress  $\sigma_{rn}$  at the radius  $r_n$  resulting from the bore pressure  $p$  is given by Equation (13a) with  $K$  replacing  $k_n$ ,  $p$  replacing  $p_{n-1}$ ,  $r_N$  replacing  $r_n$ ,  $r_n$  replacing  $r$ , and  $p_n = p_N = 0$ .  $\sigma_{rn}$  becomes:

$$\sigma_{rn} = \frac{p}{K^2 - 1} (1 - k_{n+1}^2 k_{n+2}^2 \dots k_N^2) \quad (31)$$

The pressure  $p_n$  is the sum of  $q_n$  and  $(-\sigma_{rn})$ . Therefore,

$$q_n = p_n - (-\sigma_{rn}) \quad (32)$$

The interference as manufactured,  $\Delta_n$  at  $r_n$ , is given by

$$\frac{\Delta_n}{r_n} = \frac{-u_n(r_n)}{r_n} + \frac{u_{n+1}(r_n)}{r_n} \quad (33)$$

where

$u_n(r_n)$  = radial deformation at  $r_n$  of cylinder  $N$  due to the residual pressure  $q_n$  at  $r_n$  and the residual pressure  $q_{n-1}$  at  $r_{n-1}$ .

and

$u_{n+1}(r_n)$  = radial deformation at  $r_n$  of cylinder  $n+1$  due to the residual pressure  $q_n$  at  $r_n$  and the residual pressure  $q_{n+1}$  at  $r_{n+1}$ .

Substituting the Expressions (32) for  $q_n$  into Expressions (14a) for the  $u_n$  and substituting the results into Equation (33), we find that  $\Delta_n/r_n$  reduces to:

$$\frac{\Delta_n}{r_n} = \frac{2p}{NE} \quad (34)$$

The result  $p/2S$  given by Equation (30) is plotted in Figure 43 for various  $N$ . The limit curve is given by

$$\left(\frac{p}{2S}\right)_{\text{limit}} = \frac{K^2 - 1}{K^2} \quad (35)$$

at which limit the minimum shear stress becomes equal to  $-S$  at the bore in the inner cylinder.

Figure 43 has been obtained under the assumption that  $\frac{\sigma_\theta - \sigma_r}{2}$  always gives the maximum shear stress. As pointed out by Berman, the maximum shear stress in a closed-end container\* is given by  $\frac{\sigma_z - \sigma_r}{2}$  when  $\sigma_z > \sigma_\theta$ . (42) Therefore, it is important to know the limit to  $\frac{p}{2S}$  for which  $\sigma_z$  becomes equal to  $\sigma_\theta$ .  $\sigma_z$  is given by

\*Containers for hydrostatic extrusion generally are not closed-end containers. The effect of axial stress is included here for completeness.